

Ultra-Dense LEO Satellite-Based Communication Systems: A Novel Modeling Technique

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ABSTRACT

Low Earth orbit (LEO) satellite plays an indispensable role in the equal access network because of its low latency, large capacity, and seamless global coverage. For such an unprecedented extensive irregular system, stochastic geometry (SG) is a suitable research method. The SG model can not only cope with the increasing network scale, but also accurately analyze and estimate the network's performance. Several standard satellite distribution models and satellite-ground channel models are investigated in this article. System-level metrics such as coverage probability and their intermediates are introduced in the non-technical description. Then the influence of gateway density, and the number and height of satellites on latency and coverage probability is studied. Finally, this article presents the possible challenges and corresponding solutions for the SG-based LEO satellite system analysis.

INTRODUCTION

In the last few years, we have been witnessing a boom in the industry of low Earth orbit (LEO) satellite networks. Constellations such as Starlink, Telesat, and OneWeb are planning to deploy or have already deployed thousands of satellites [1]. With relatively larger capacity and lower latency, LEO satellites provide an affordable solution for seamless network coverage, which enables access equality. However, a strong and tractable mathematical framework is still needed to analyze the performance of LEO satellite-based communication systems. In particular, for a massive satellite constellation deployed at different altitudes from the Earth, the analysis will enable us to understand the required numbers and altitudes of satellites to maximize system performance. In addition, it will enable us to learn how many Earth stations are needed in order to enable seamless operation of the LEO satellite communication system.

Based on the above description, the structure of this article is as follows. First, we describe the advantages and disadvantages of LEO satellites in more detail. Then we give the motivation for applying SG to LEO satellite networks. Based on SG, this article goes over the analysis framework of some recent literature and gives a non-technical description of models of satellite locations distri-

bution, channel model, typical parameters, and main performance metrics. In addition, a simulation setup is proposed to highlight the potential gains of using SG in modeling and analyzing LEO satellite communication systems with useful system-level insights. Finally, we predict future scenarios for the challenges of applying SG in satellite networks.

ADVANTAGES AND CHALLENGES OF LEO SATELLITE SYSTEM

In this section, the advantages and challenges of the LEO satellite system are analyzed by comparing it to the geosynchronous Earth orbit (GEO) satellite system and the terrestrial system [2]. The radar diagram in the lower left corner of Fig. 1 visually shows the qualitative comparison points.

The LEO satellite system can provide seamless coverage of many regions around the globe that suffer from partial or no connectivity, and has incomparable advantages in access equality. Despite the enormous coverage of a single GEO satellite, the system has blind areas at the poles. The coverage of terrestrial systems is limited by terrain. In remote areas that lack infrastructure, existing ground networks are challenged to provide coverage because the cost of extending fiber optics to these areas is expensive.

The LEO satellite system has relatively larger capacity density and lower latency. In the same region, LEO satellite systems provide much more capacity than GEO satellite systems, but less than terrestrial systems. The capacity is sufficient to support high rate communications in rural and remote areas, but not enough to meet the needs of densely populated cities. The latency of LEO satellites ranges from several milliseconds to tens of milliseconds, which can meet the needs of most application scenarios. Application scenarios requiring high real-time performance, such as the Internet of Vehicles, are still highly dependent on base stations (BSs). The long communication distance of the GEO satellite system leads to a much longer delay than the others.

However, LEO satellite systems face significant challenges in terms of cost and frequency coordination. A large LEO satellite system requires high construction and maintenance costs. Fortunately, aerospace manufacturers have brought down launch costs, and streamlined production is bringing down the cost of producing satellites. In terms of frequency licensing, the International Telecom-

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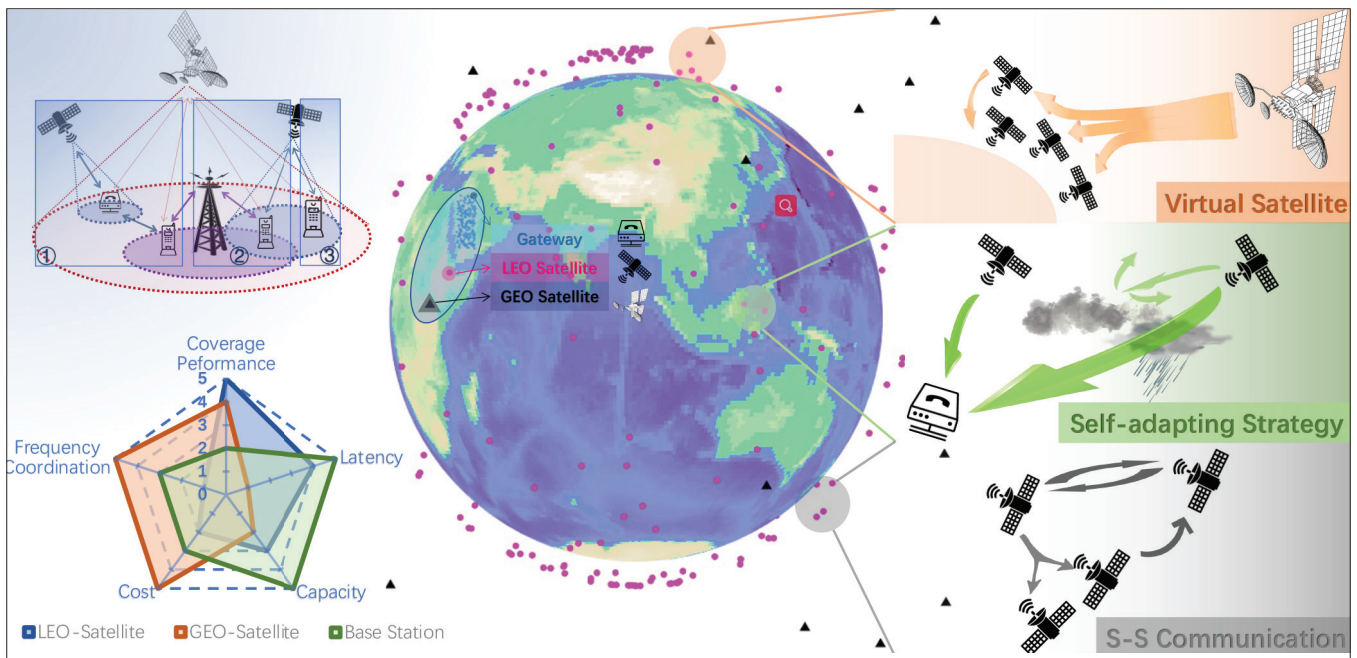


FIGURE 1. System-level metric comparison of three systems and three future scenarios.

munication Union (ITU) gives frequency priority to GEO satellites. Because of the first-claim, first-possession principle, once these huge constellations are built, it is not easy to allocate dedicated spectrum to the new constellations. There is also some channel overlap between LEO satellite systems and 5G ground BS systems, which further complicates the challenges of spectrum resource allocation problems.

MOTIVATIONS OF APPLYING SG IN LEO SATELLITE NETWORKS

As a powerful mathematical tool, SG has been widely used in communication networks. The advantages of SG-based modeling and analysis in LEO satellite system is given as follows.

The existing satellite analysis methods are mainly to design the specific constellation. However, for such a large system, fine-grained modeling of each LEO satellite would take a lot of effort. It is far more tractable to do SG-based system-level analysis than to define and study the behavior of each satellite. SG modeling does not need to rely on orbital metrics and the shape of a specific constellation. A universal and flexible random network is what the explosive growth of dynamic networks requires.

For satellite systems, more satellites and lower altitudes mean larger density. In this case, interference will be an important factor affecting system performance. However, apart from SG, there is still a lack of practical analytical tools to simulate the interference caused by satellites.

In the existing non-SG models, satellites are often deployed in regular circular or hexagonal cells with the same coverage [3]. However, the distribution of satellites depends on latitude, and the coverage of satellites varies significantly in most cases. Therefore, the results of traditional analysis methods may greatly differ from the actual situation. SG is suitable for modeling and analyzing irregular topological networks [4]. In recent studies, various types of point processes used in LEO satellite networks allow multiple methods to simulate the distribution.

Applying SG requires the assumption that satellites are independent and therefore can only be used to study the lower bounds of system performance. However, it has been proved that the lower bound of coverage probability and average achievable rate in the SG model is as tight as the upper bound of the regular mesh model [5]. For the LEO satellite system, the performance of the SG model closely matches with that of the deterministic constellation in coverage probability and average achievable rate [6].

SG-BASED ANALYTICAL FRAMEWORK

In this section, we go over the analytical framework based on SG in recent works. Table 1 provides a brief summary of these frameworks. Before introducing the model, we first introduce some definitions in astronomy:

- **Latitude:** Latitude is the line plane angle between the normal line on the Earth and the equatorial plane. It ranges from 0° at the equator to 90° at the poles.
- **Zenith Angle:** Zenith angle is the angle between the incident signal and the direction of the local zenith (the geographical vertical).
- **Horizon:** The horizon is a two-dimensional plane tangent to the Earth through the perception point, which can be the location of the gateway (GW) or the user. From the GW's perspective, any satellite below the horizon is beyond line of sight since the Earth acts as a blockage.

MODELS OF SATELLITE LOCATIONS DISTRIBUTION

In most of the existing literature, it is assumed that satellites, as well as users, satellite GWs, and BSs, are independent and uniformly distributed. The following sections discuss the three-point process distribution of satellite networks.

For Poisson point process (PPP), the number of points in a predefined area is Poisson distributed while their locations are uniformly distributed within this area. As one of the most frequently

References	Scenarios	Distribution of satellites	Small-scale fading	System metrics	System types
[6, 9, 13]	Direct communication	Non-homogeneous PPP BPP	Non-fading Rayleigh fading Rician fading	Coverage probability Average achievable rate	Noise-limited system Interference-limited system Generic system
[7, 11]	Hybrid network	PPP	Rayleigh fading	Coverage probability	Ideal system Noise-limited system Generic system
[8, 10]	Multi-tier system GW-relayed	BPP	Shadowed-Rician fading	Coverage probability	Noise-limited system

TABLE 1. Models used in different literature.

used point processes, PPP can fit the ground network well in performance analysis such as coverage probability and achievable data rate. Therefore, the positions of the LEO satellite are modeled as a PPP on spherical surfaces of fixed height [7]. One of the most significant advantages of PPP is that it can predict the probability of a certain number of points in a spherical region with a specific area measure.

Although PPP has a strong practicality, it is not the best choice for modeling a finite area network with limited nodes since the positions of satellites are on spherical surfaces, and the number of satellites is fixed for deterministic constellations. Therefore, modeling the position of the satellite as a binomial point process (BPP) is an effective solution [6, 8]. Specifically, it places a fixed number of points on the fixed height sphere. The zenith (elevation) and azimuth angle are uniformly distributed.

Non-homogeneous Poisson point process (NPPP) improves the PPP model from another perspective. Since the distribution of deterministic satellite constellations is not uniform at different latitudes (i.e., the number of satellites at the limit of inclination is actually greater than that in the equatorial region), the number of effective satellites per latitude in NPPP is used to compensate for the uneven density [9].

Corresponding to the upper left part of Fig. 1, there are three typical scenarios: GW-relayed system, terrestrial station, and LEO satellite hybrid network and satellite to user direct communication. The GW can provide directional transmission to the satellite. By deploying a GW as a relay, satellite coverage can be significantly improved [10]. The satellite network can also assist the ground network to further increase system capacity and provide communications in emergency situations. Satellites can also communicate directly with users, which of course requires special user equipment to enable direct communication. Since the coverage radius of the GW or BS is negligible compared to the surface area of the Earth, GWs and BSs can be regarded as points located in a large two-dimensional plane, and PPP is suitable to model their locations. As shown in the middle part of Fig. 1, GEO and LEO satellites form two independent BPPs distributed on different spheres, and the ground GWs form a PPP.

CHANNEL MODEL

The linear expression of the received power is $\rho \eta Cr^{-\alpha}$, where ρ is used to describe the transmitted power and antenna gain, η is used to describe large-scale fading, C is used to describe small-scale

fading, and α is called the path loss exponent.

Take the downlink transmission as an example. There are two main models of signal power transmission. In the first model, satellites have the same gain in all directions, which is called the equivalent isotropically radiated power (EIRP) criterion [7]. In the second model, satellites have a larger gain in a particular direction realized by beamforming (BF) technology [11]. The GW or user can receive main-lobe power in this model, while the power from the interfering source tends to be side-lobe, which can improve the signal-to-interference ratio (SIR) of the system.

The path loss exponent is a number between 2 and 4 usually. Consider that a satellite-to-ground link propagates in free space; in most cases, $\alpha = 2$ is satisfied for open areas. However, when the ground clutter tier reflects and absorbs the power, the multi-path effect cannot be ignored, resulting in $\alpha = 4$. An intuitive result is that when $\alpha = 2$, the interference becomes the main factor limiting system performance, and when $\alpha = 4$, the influence of noise will be far more than that of interference.

Large-scale fading is usually modeled as a log-normal shadowing to describe additional gain. It is inversely proportional to the square of carrier frequency [6–8]. Furthermore, some of the literature takes the attenuation of air absorption caused by the resonance of water vapor into consideration [8, 11], which is called rain attenuation. It changes with geographical location, such as desert and rain forest.

Small-scale fading is a random variable that denotes the channel fading power gains. In ideal free space propagation, small-scale fading does not occur. Non-fading (absence of fading) helps to investigate the upper bound on system performance. On the contrary, Rayleigh fading helps investigate the lower bound, which happens when serious multi-path distortion occurs. Rician fading is suitable for the general situation and is widely applied to make accurate performance estimates. Shadowed-Rician (SR) fading is a Rician fading channel with fluctuating line of sight (LoS) components. It is the most accurate channel fading model, especially for space-to-ground links [8]. There are multiple available channel models in the literature such as Loo distributions and Nakagami- m fading. However, the most commonly used models in the SG literature are the SR model, Rayleigh, and Rician fading.

Handling non-line-of-sight (NLoS) interference is one of the keys to the accuracy of a channel model. Satellites below the horizon may not provide reliable service, but the diffraction signal may still interfere with the user. At the same time,

When LEO satellites rely on a ground GW as a relay, a user is covered by a satellite when both satellite-GW and GW-user links achieve the SINR requirements. In the terrestrial BS and LEO satellite hybrid network scenario, satellites and BSs will independently establish the network.

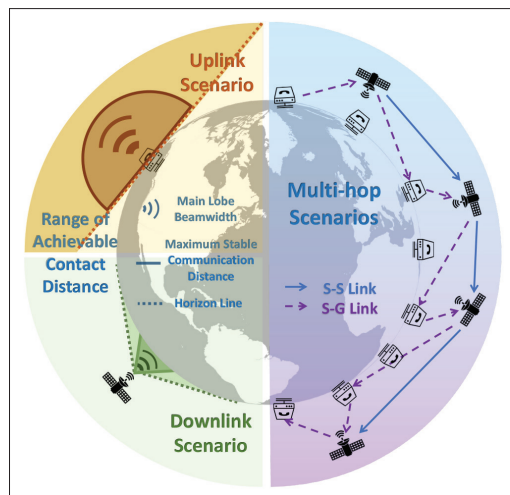


FIGURE 2. LEO satellite system structure diagram.

the satellite may not establish an LoS link due to obstacles from the ground. The former is often regarded as zero or some small constant rather than a random variable because the signal is relatively weak. The latter can be described by SR fading. It has been proved by simulation that SR fading can accurately simulate the channel model under NLoS interference [12]. In addition, multiplying by a factor representing the loss of redundant paths in large-scale fading is also an effective solution. This factor is modeled as a log-normal mixture random variable with two components, and the mixing ratio is given by the probabilities of LoS and NLoS [7]. Notice that the probability that an LoS link is established is only determined by the zenith angle.

CONTACT DISTANCE AND SATELLITE AVAILABILITY

Contact distance is the distance between the user/GW and the associated (service-providing) satellite. References [6, 10] provide two expressions of contact distance distribution for BPP. Based on the strongest average received power association strategy, the contact distance is the distance from the user to the nearest satellite at a given altitude. As a random variable, the complementary cumulative distribution function (CCDF) of contact distance at d_0 is equal to the probability that there are no satellites in the spherical cap at the intersection of spherical surface over which satellites are distributed and the cone with side length d_0 centered at the location of the user/GW region closer to the user. Contact angle is defined as the minimum zenith angle between the user and the satellite. As shown in [7], the contact angle can be regarded as the representation of the contact distance in spherical coordinates.

From spatial distribution perspectives, satellites, GWs, and users are all located on spherical surfaces. The upper left region of Fig. 2 shows the range of possible locations of a serving LEO satellite for a given location of the user/GW in the downlink. This range is mainly defined, as explained earlier, by the horizon, below which no LoS satellites are available [11]. When BF is used, the main-lobe beamwidth is also one of the limitations [13]. The range of achievable contact distances is the intersection region above the horizon and in the beam's main-lobe width.

Nearest neighbor is used to describe the shortest distance from a selected satellite to another. It is an important part of multihop network analysis. The difference between nearest neighbor and contact distance is that the user/GW does not belong to the satellite point process, while the selected satellite in the nearest neighbor does [10].

When the satellites are deployed at different altitudes with different values of transmission power, the concept of *association probability* needs to be introduced [14]. The association probability with a given tier is the probability that the nearest satellite from this tier provides higher average received power than that of the nearest satellites from all other tiers.

Satellite availability means that at least one satellite is within the visible range of the user/GW. In small, low-altitude constellations, it is common to have no usable satellites above the horizon. In the existing literature, there are three methods to deal with the satellite availability problem:

1. Reduce the probability of having no usable satellites below an acceptable threshold by increasing satellite density [10].
2. Enhance the probability of having available satellites by increasing the height and transmitting power simultaneously [11, 13].
3. Calculate the probability of no available LoS satellite and set signal-to-interference-plus-noise ratio (SINR) to zero if no LoS satellites are available [6].

The first two proposals provide guidelines for constellation designers, and the last proposal provides the most accurate estimation of system performance for a designed constellation.

COVERAGE PROBABILITY AND AVERAGE ACHIEVABLE RATE

In this section, we study the application of SG in system performance analysis. Three related definitions for measuring system performance are:

- The **SINR** is used to measure the communication quality of the system.
- **Coverage probability** is defined as the probability that the SINR is larger than a predetermined acceptable threshold. It represents the probability that the system can provide reliable connections.
- **Average achievable rate** is the ergodic capacity from the Shannon-Hartley theorem over a fading communication link.

After the distance to the associated satellite is determined by the contact distance, the sum of the power of other LoS satellites is the interference. SINR can be obtained; then the coverage probability and average achievable rate are calculated. When a LEO satellite adopts multiple orthogonal bands, the average achievable rate should be divided by the number of orthogonal bands.

Coverage probabilities in different application scenarios are expressed differently. When LEO satellites rely on a ground GW as a relay, a user is covered by a satellite when both satellite-GW and GW-user links achieve the SINR requirements. In the terrestrial BS and LEO satellite hybrid network scenario, satellites and BSs will independently establish the network. The user is considered in outage when the SINR of either the user-GW link or the GW-satellite link is below the threshold.

Coverage probabilities in different system types are highlighted in the literature. In an ideal system, the signal will not be affected by any interference and noise in transmission. As long as there is a satellite in service, the user can always acquire a successful transmission. Such systems are used to study the upper bounds of performance and satellite availability. A noise-limited system occurs in an environment where the noise power is much stronger than the interference power. The interference power is often regarded as zero or a small constant. It happens when the interference comes from a diffraction signal. In an interference-limited system, interference power is dominant. Using orthogonal channels will significantly reduce the co-channel interference and improve the coverage probability. However, each satellite can use only a portion of the system's bandwidth, which reduces capacity and data rates. Therefore, there is an optimal number of orthogonal channels in interference-limited systems to maximize the average achievable rate [6]. A generic system takes both interference and noise power into account.

Coverage probabilities and average achievable rate under different types of small-scale fading have completely different mathematical expressions. Coverage probability and average achievable rate under non-fading and Rayleigh in the generic system are provided in [6]. In Rician fading and SR fading, the associated satellite signal is transmitted in the direct path, while the interfering satellite signals are transmitted through the reflection path. In this case, the system is noise-limited. In [8], the coverage probability under SR fading in a noise-limited system is obtained.

Finally, no matter what kind of small-scale fading is adopted, the Laplace transform of interference is involved in the calculation of coverage probability and achievable rate [6].

SG-BASED SIMULATION SETUP

Suppose two GWs at opposite ends of the Earth, with zenith angles 0 and π , need to communicate. Satellites are modeled as a BPP on a sphere concentric with the Earth. The signals of the inter-satellite link and the satellite-ground link travel at the speed of light. As shown in Fig. 2, the blue arrow represents satellite-satellite (S-S) link, and the purple arrow represents satellite-gateway (S-G) link. GWs can send messages to the associated satellites, while satellites can communicate with the GWs in the range of LoS. We designed two scenarios [15]:

1. In the *no inter-satellite links scenario*, the satellite uses GWs to forward data to other satellites until the message is delivered to its destination.
2. Instead of requiring a GW relay, the satellite sends a message directly to the next hop in the *inter-satellite links scenario*. Examples of both scenarios are shown in Fig. 2.

In Fig. 3, it can be seen that when the number of satellites is insufficient, the transmission latency decreases with the increase of altitude. This is because when a satellite is looking for the next hop, it can only try to find satellites in LoS that might transmit signals. As it rises in altitude, the satellite's range of view expands, allowing it to choose which of several satellites is in a better position. However, for enough satellites, with the

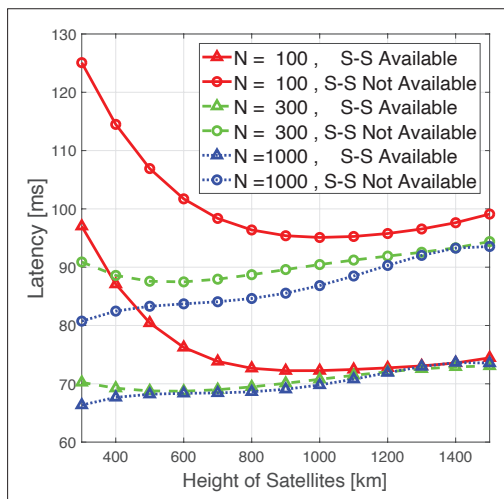


FIGURE 3. Average latency of satellite systems at different altitudes and scales.

increase of height, the distances of both inter-satellite link and satellite-ground link will increase.

Overall, when S-S communication is not available, the delay increases by about 30 percent compared to the case when S-S communication is available. When the number of satellites is increased from 100 to 300, the system delay can be significantly reduced. From 300 to 1000 satellites, performance no longer improves much. In particular, when S-S communication is not available, the curves of 300 satellites and 1000 satellites overlap, and adding additional satellites will not bring additional benefits in terms of delay. As the altitude increases, the two considered scenarios tend to have different latency values. The trends of the three curves when S-S communication is available tend to be consistent faster. It suggests that increasing altitude, increasing constellation size, and allowing interaction between satellites lead to more satellites within a satellite's reach. When there are enough satellites in the range, the performance improvement of these methods is limited, and the trend of the curves tends to be consistent.

In Figs. 4 and 5, we analyze the influence of satellite number, GW density, and constellation altitude on system coverage probability. Satellites follow BPP, and GWs follow PPP. We assume that the transmitting power of the satellite is 15 dBw. The total coverage probability of the system is the product of the coverage probability of two links:

1. Satellite-GW link
2. GW-user link

The satellite-GW link and GW-user link follow the same channel fading, large-scale fading is 0 dB in LoS range, small-scale fading follows $SR(1.29, 0.158, 19.4)$ [8], and the path loss exponent is 2. In particular, the large-scale fading of 0 dB means that the effects of wavelength, non-directional antenna gain, and rain attenuation are normalized. Assume that satellites and gateways use the same frequency band, and co-channel interference exists. When calculating the coverage probability, the threshold of SINR is -10 dB.

In Fig. 4, the satellites have an altitude of 1000 km, that is, they are distributed on a spherical surface with a radius of 7371 km (radius of the Earth is 6371 km). When satellites are insufficient, the system lacks available satellites to provide cover-

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This article investigates the application of SG to simulating and analyzing LEO satellite communications systems, introduces several satellite distributions and analyzes the similarities and differences of channel models in the literature. From a non-technical perspective, we describe coverage probability in different scenarios, average achievable rate, and contact distance.

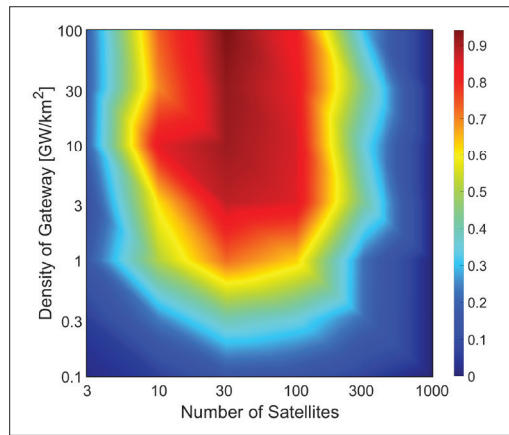


FIGURE 4. Influence of satellite number and GW density on coverage probability.

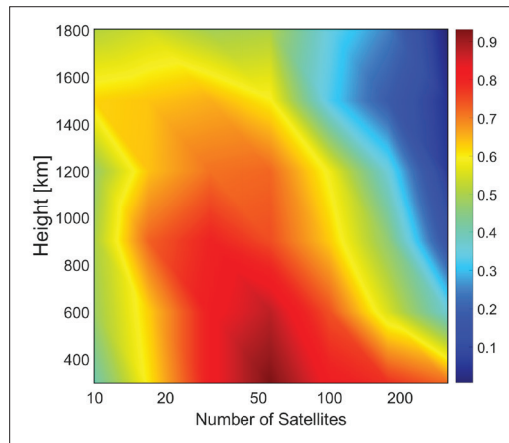


FIGURE 5. Influence of satellite number and constellation height on coverage probability.

age, or the associated satellites are too far away. Conversely, too many satellites can cause significant interference. Therefore, with the increase of the number of satellites, the coverage probability shows a trend of first increasing and then decreasing. In addition, at an altitude of 1000 km, the optimal number of satellites for coverage probability is about 30. The optimal number of satellites is independent of the density of GWs. Unlike satellites, increasing the density of GWs in a reasonable range will improve the coverage performance of the system, and the GW subsystem shows the characteristics of a noise-limited system.

In Fig. 5, the density of GWs is fixed at $3/\text{km}^2$. The increase of constellation height will reduce the coverage probability. There is an optimal number of satellites corresponding to altitude. With the increase of constellation height, the number of satellites corresponding to the optimal coverage decreases.

CHALLENGES AND FUTURE SCENARIOS

Although SG has many advantages in system performance analysis, the traditional SG analysis framework has some limitations, and there are many challenges in applying SG to more complex and sophisticated satellite systems in the future. Therefore, in the right part of Fig. 1, we envision several future scenarios. These scenarios can also be modeled, analyzed, and evaluated using SG-based models as described in this article.

S-S communication: In the previous analysis, we assumed that satellites were independent of each other and analyzed the lower bound of the system performance. However, in future satellite systems, there will be more S-S communications, which will significantly improve the system's performance. Through interaction, satellites can learn more about the environment, such as the number of nearby satellites and the frequency bands they use. Communicating with satellites in front of a predetermined trajectory allows them to predict future communication environments. The problem of insufficient capacity and achievable data rate can be effectively alleviated by smartly designing association and handover techniques between satellites. Conversely, with the application of orthogonal channels, it is possible to have a high SINR even if the user/GW is associated with a non-optimal satellite after the handover.

Self-adapting to the environment: The second challenge faced by traditional SG analysis frameworks is that in reality, satellites do not remain inactive when communications fail in harsh environments. While facing different astronomical and geographical conditions, actively perceiving the environment and selecting proper adaptive strategies are important. For example, the satellite can always point the main-lobe to the area with the greatest demand for services, greatly improving local coverage.

Virtual satellite: In addition to S-S communications, there may be direct cooperation between satellites. Similar to the idea of virtual hosting, multiple satellites with the same mission can be integrated into a virtual satellite. The virtual satellite can decide the number of orthogonal channels and allocate specific frequencies internally. Furthermore, LEO satellites can be used as the auxiliary of GEO satellites to complement the coverage of regions that a single GEO cannot meet independently.

In addition, the concept of software-defined networking plays an important role in satellite constellation management. Network functions virtualization enabled and service function chain embedded satellite systems are promising development directions.

CONCLUSIONS

This article investigates the application of SG to simulating and analyzing LEO satellite communications systems, introduces several satellite distributions, and analyzes the similarities and differences of channel models in the literature. From a non-technical perspective, we describe coverage probability in different scenarios, average achievable rate, and contact distance. This article analyzes several factors affecting latency and coverage probability using SG-based models and gives the following conclusions. Increasing the number of satellites and the height of the constellation within a specific range can effectively reduce the latency. The ability of the satellites to communicate with each other has a considerable effect on reducing the value of latency by 30 percent. This indicates that the recent technological advances in enabling S-S communications will significantly widen the set of applications that can rely on LEO satellite communications.

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